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PRINT DATE: 12/19/95

FAILURE MODES EFFECTS ANALYSIS (FMEA) - NON-CIL HARDWARE NUMBER:05-6Q-2112 -X

SUBSYSTEM NAME: EPD&C - DISPLAYS & CONTROLS

REVISION: 1

12/18/95

PART DATA

PART NAME

VENDOR NAME

PART NUMBER

VENDOR NUMBER

LRU

:PANEL 013

V070-730393

SRU :CIRCUIT BREAKER

MC454-0026-2050

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS: CIRCUIT BREAKER

REFERENCE DESIGNATORS:

33V73A1CB1

33V73A1CB9

QUANTITY OF LIKE ITEMS:

TWO

FUNCTION:

PROVIDES CIRCUIT OVERLOAD PROTECTION FOR ESSENTIAL BUSSES 1BC AND 2CA. POWER TO MASTER ALARM LIGHTS VIA RELAY AND/OR TOGGLE SWITCH AND TO C&W ANNUNCIATOR CIRCUITRY AND ELECTRONICS UNIT.

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FAILURE MODES EFFECTS ANALYSIS FMEA - NON-CIL FAILURE MODE

NUMBER: 05-6Q-2112-01

REVISION#: 1

12/18/95

SUBSYSTEM NAME: EPD&C - DISPLAYS & CONTROLS

LRU: PANEL 013

CRITICALITY OF THIS

ITEM NAME: CIRCUIT BREAKER

FAILURE MODE: 1R3

FAILURE MODE:

LOSS OF OUTPUT, FAILS OPEN. INADVERTENTLY OPENS.

MISSION PHASE:

PL PRE-LAUNCH LO LIFT-OFF OO ON-ORBIT

DO DE-ORBIT

LS LANDING/SAFING

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102 COLUMBIA

103 DISCOVERY 104 ATLANTIS

105 END€AVOUR

CAUSE:

PIECE PART STRUCTURAL FAILURE, CONTAMINATION, MECHANICAL AND/OR THERMAL STRESS AND VIBRATION.

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) PASS

C) PASS

PASS/FAIL RATIONALE:

A)

B)

C)

CORRECTING ACTION: MANUAL

CORRECTING ACTION DESCRIPTION:

THE CREW MUST CONTINUALLY MONITOR FAULT SUMMARY MESSAGES ON THE DISPLAY UNIT (CRT) FOR LOSS OF THE ASSOCIATED ESSENTIAL BUS UNTIL POWER IS PAGE: 3 PRINT DATE: 12/19/95

FAILURE MODES EFFECTS ANALYSIS (FMEA) — NON-CIL FAILURE MODE NUMBER: 05-6Q-2112-01

RESTORED TO THE AFFECTED POWER SUPPLY THROUGH AN IFM PROCEDURE, THE REMAINING POWER SUPPLY IS REDUNDANTLY POWERED THROUGH AN IFM PROCEDURE OR THE ASSOCIATED FUEL CELL IS PLACED IN STANDBY.

- FAILURE EFFECTS -

(A) SUBSYSTEM:

LOSS OF CAPABILITY TO CONDUCT POWER.

(B) INTERFACING SUBSYSTEM(S):

LOSS OF POWER TO CAUTION AND WARNING A OR C&W B.

(C) MISSION:

NO EFFECT.

(D) CREW, VEHICLE, AND ELEMENT(S):

NO EFFECT.

(E) FUNCTIONAL CRITICALITY EFFECTS:

EACH POWER SUPPLY IS CONNECTED TO A DIFFERENT ESSENTIAL BUS THROUGH CAW CIRCUIT BREAKERS. LOSS OF ONE CWE POWER SUPPLY WILL TRIGGER A CAW ALARM. THE CREW ACKNOWLEDGES THE ALARM BY DEPRESSING THE MASTER ALARM RESET SWITCH. IF THE REMAINING POWER SUPPLY, OR ITS ESSENTIAL BUS FAILS, THE RESULT WILL BE THE LOSS OF ALL CAW AURAL AND VISUAL ALARMS. THE ESSENTIAL BUS FAILURE WILL SIMULTANEOUSLY INTERRUPT FUEL CELL COOLANT PUMP OPERATION, CREATING A TIME CRITICAL EMERGENCY CONDITION. THE CREW MUST TAKE REMEDIAL ACTION WITHIN 9 MINUTES OF ESSENTIAL BUS FAILURE TO AVOID A CATASTROPHIC FUEL CELL FAILURE.

DESIGN CRITICALITY (PRIOR TO DOWNGRADE, DESCRIBED IN (F)): 1R2

(F) RATIONALE FOR CRITICALITY DOWNGRADE:

AFTER THE FIRST FAILURE THE CREW WILL PERFORM AN IFM TO RESTORE POWER TO THE AFFECTED C&W OR TO REDUNDANTLY POWER THE REMAINING C&W UNIT AND PRECLUDE A SINGLE FAILURE (ESSENTIAL BUS LOSS) FROM RESULTING IN AN UNANNUNCIATED TIME CRITICAL CATASTROPHIC FUEL CELL FAILURE.

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FAILURE MODES EFFECTS ANALYSIS (FMEA) - NON-CIL FAILURE MODE

NUMBER: 05-6Q-2112-01

- APPROVALS -

EDITORIALLY APPROVED

: RI : JSC 9 mil 12/19/95

EDITORIALLY APPROVED
TECHNICAL APPROVAL

: APPROVAL FORM

95-CIL-003-RI